Primljen / Received: 15.12.2012. Ispravljen / Corrected: 18.1.2013. Prihvaćen / Accepted: 21.2.2013. Dostupno online / Available online: 10.5.2013.

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Target acceleration in multimodal pushover method for R/C frames

Preliminary note

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Target acceleration in multimodal pushover method for R/C frames

The procedure of searching the target ground acceleration as a measure of the lowest seismic resistance is presented in the paper. This objective is achieved by the multimodal pushover analysis based on the envelope principle. The pushover method founded on the linear combination of modes (L), and on mode combinations as a square root of the sum of the squares (SRSS), is presented. Examples of 5-storey and 9-storey R/C frames point to a highly significant influence of higher modes. The failure ground acceleration in multimodal pushover analysis is significantly lower, and even several times lower, compared to the single mode acceleration.

Key words:

pushover method, combining of modes, R/C frame, target acceleration, general failure force vector

Prethodno priopćenje

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Ciljano ubrzanje u višemodalnoj metodi naguravanja A/B okvira

U radu je prikazan postupak traženja ciljanog ubrzanja tla kao mjera najniže potresne otpornosti. Dobivanje ciljanog ubrzanja postiže se višemodalnim naguravanjem po načelu anvelope. Prikazano je naguravanje linearnom kombinacijom modova (L) i kombinacijom modova kao kvadratnog korijena iz sume kvadrata (SRSS). Na konkretnim primjerima 5-katnog i 9-katnog a/b okvira pokazuje se vrlo značajan utjecaj viših modova. Slomno ubrzanje tla u višemodalnom naguravanju značajno je manje, pa i višestruko manje od ubrzanja s jednim modom.

Ključne riječi:

metoda naguravanja, kombiniranje modova, a/b okvir, ciljano ubrzanje, opći slomni vektor

Vorherige Mitteilung

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Zielbeschleunigung in der multimodalen Pushover Analyse für Stahlbetonrahmen

In der vorliegenden Arbeit ist die Berechnung der Zielbodenbeschleunigung als Maß der niedrigsten Erdbebenbeständigkeit dargestellt. Die Zielbeschleunigung wird, auf dem Hüllkurven-Prinzip beruhend, durch eine multimodale Pushover Analyse ermittelt. Der Lastvektor ist sowohl als lineare Kombination (L) als auch in der Form der Quadratwurzel aus der Summe der Quadrate (SRSS) der Schwingformen beschrieben. Anschauliche Beispiele eines 5-stöckigen und eines 9-stöckigen Stahlbetonrahmens weisen auf den bedeutenden Einfluss höherer Schwingformen. Die kritische Bodenbeschleunigung bei der multimodalen Pushover Analyse ist bedeutend und sogar mehrfach kleiner als bei der ausschließlichen Berücksichtigung der ersten Eigenschwingung.

Schlüsselwörter:

Pushover Methode, Kombination von Schwingformen, Stahlbetonrahmen, Zielbeschleunigung, allgemeiner kritischer Lastvektor

1. Introduction

The pushover method is an efficient procedure for the nonlinear analysis of earthquake resistance. In combination with the response spectrum for a single-degree-of-freedom system (SDOF), the method enables evaluation of the bearing capacity and strain of structures subjected to earthquake action. The method has been incorporated in various international codes for structural analysis, and is recommended by research institutions [1-3]. The development of the procedure adopted in EN 1998 was initiated three decades ago [4, 5], and various improvements and modifications, such as those proposed in papers [6, 7], are still being made [8]. The pushover method based on the distribution of accelerations according to the first mode, and constant distribution of accelerations along the height of the structure, is dominantly used in current practice. The influence of higher modes in nonlinear structural analysis based on pushover method has been analysed in many papers [9-15], and the authors agree that the influence of higher modes is considerable.

A multimodal pushover method, aimed at including higher modes so as to meet the envelope principle and define the least resistance, is presented in this paper. A concrete application of the method is possible for a specific elastic spectrum. The type 1 spectrum for the soil type A according to EN 1998 is used in examples presented in the paper. The bearing capacity curves are determined by pushover method separately for each mode, and are then converted into the ADRS format (acceleration displacement response spectrum) for the spectrum selected in advance. The form of modes a d period sizes are determined using the linear elastic model.

The method can be applied using the following set of steps:

- The procedure starts by selection of a concrete elastic spectrum.
- Then the hypothetic peak soil acceleration value is selected, usually the lowest one of all modes.
- An appropriate spectral acceleration and spectral load level is determined for each mode separately, based on the bearing capacity curve, taking into account the target displacement according to EN 1998. The form of load vector corresponds to the form of an appropriate mode.
- In addition, a general failure load vector is formed according to some possible load combinations for real modes. The following possible mode combinations are presented in the paper: linear (L) combination, and combination of the square root of the sum of the squares (SRSS).
- The bearing capacity, i.e. the design peak capacity, is determined for the load formed in this way, or general failure vector, by means of the non-linear analysis of the structure using the pushover method.

The task is considered solved when the load equality is obtained, within the accuracy set in advance, from the general failure vector and the design limit load for the assumed soil acceleration. The procedure conducted in several steps leads to a rapid solution for the selected different hypothetic acceleration and the corresponding vectors. The result obtained is the target acceleration which is the lowest acceleration that leads to the design limit state. Based on the examples of linear (L) combination of modes, and combination of the square root of the sum of the squares (SRSS), it can be concluded that the target acceleration is always lower that the limit acceleration for each mode regarded separately. This fact is the initial step in the method. The procedure presents properties of the envelope principle.

2. Determination of target acceleration

As previously indicated, the target acceleration seeking procedure starts by determination of the shape vector $\phi_{r'}$ corresponding periods T_r and participating masses $m_{er'}$ based on the linearly elastic analysis. The load vector F_i is determined separately for each *i*-th mode through the total action intensity factor p_r as the product of the mass matrix **M** and the vector ϕ_i

$$\mathbf{F}_i = \boldsymbol{\rho}_i \, \mathbf{M} \, \phi_i \tag{1}$$

Load bearing capacity curves were calculated by the numerical model of stability and bearing capacity of spatial linear structures, taking into account material and geometric nonlinearities [16], with a monotonous increase of the load vector **F**i. Failure-generating transverse forces in the crosssection were determined in this way at the basis of the calculation model separately for each mode.

When looking for the target acceleration, each attempt starts with an assumed target ground acceleration value a_{gr} . The ADRS curve is established for the assumed a_{gr} elastic spectrum type (type 1 or type 2), type of soil (soil parameter S), and damping correction factor (h=1 for z=5% viscous damping). Bearing capacity curves are converted into the ADRS format, separately for each mode.

The spectral acceleration $a_{s,i}$ is determined separately for each mode, according to EN 1998 and Annex B, so that the displacement at failure amounts to 150% of target displacement on the bearing capacity curve.

$$d_t = \frac{2}{3}d_u \tag{2}$$

The corresponding elastoplastic equivalent (bilinear forcedeformation diagram), i.e. the diagram in which the intercept of the period T_i and the related target displacement d_t falls on the ADRS curve of the assumed ground acceleration, must be determined. In other words, an equality between the ground acceleration a_{gri} obtained in the i-nth attempt and the assumed ground acceleration a_{gri} must be established, in order to define the spectral acceleration a_{sri}

The procedure for determining the corresponding bilinear forcedeformation diagram is presented in Figure 1.

The selected bilinear force-deformation diagram is presented in Figures 1.a1) and 1.a2) for the case when the period is $T_{i} \ge T_{c}$. In Figure 1.a1), the selected diagram gives greater ordinates, and in Figure 1.a2) smaller ordinates, on the ADRS curve, when compared to the ADRS curve for the assumed ground acceleration a_{α} . In Figure 1.a3), the selected diagram for the



Figure 1. Determination of spectral acceleration $a_{s,i}$ a1)-a3) for T_i \ge T_c; b1)-b3) for T_i < T_c

corresponding period T_i and the related target displacement d_t gives the intercept on the ADRS curve, which enables definition of the spectral acceleration $a_{s,r}$. A similar procedure is also conducted for the case when $T_i < T_{cr}$ i.e. when the intercept is defined according to EN 1998 – Annex B.

Once the preceding step for each vector is determined, the parameter δ_i is defined as the ratio of the obtained spectral acceleration to the assumed ground acceleration.

$$\delta_{i} = \frac{a_{s,i}}{a_{gr}} = \delta_{i} \left(a_{gr} \right) \tag{3}$$

Furthermore, possible modal combinations are also defined. As already indicated, linear (L) combination, and combination of the square root of the sum of the squares (SRSS) is monitored. The target acceleration result is obtained via repeated attempts, by varying the assumed hypothetic acceleration as an input data.

2.1. Target acceleration toward the linear (L) combination of modes

The starting hypothetical limit load ΣF_i for linear combination of modes depends on the initial acceleration selected, and can be presented as follows:

$$\sum F_{i} = \sum \pm m_{ei} a_{si} \left(a_{gr} \right) F_{fi} / |F_{fi}|$$
(4)

where the sign ± means more unfavourable effect on the typical cross-section. The typical cross-section for a RC frame example is presented via the sum of bending moments at the bottom of the lowest storey. The influence line for typical values can be used during selection of the sign.

The expression F_{fi} represents the failure force for a particular eigenvector. It is assumed in numerical procedure that the load is applied in increments. As a result, the associated limit load was obtained, i.e. the total transverse force in the cross-

section at the base of the calculation model F_L for the entire system.

The adequacy of the initial assumption is shown by the ratio of the design limit load F_{l} to the initial hypothetical load vector ΣF_{r} . As soon as their equality is obtained, within limits of the predefined design accuracy, it can be stated that the desired result has been obtained, which is the extreme limit acceleration, i.e. the target acceleration $a_{gr,t}$ corresponding to the smallest design ground acceleration.

2.2. Target acceleration according to the root combination of modes (SRSS)

The root combination, i.e. the method of the square root of the sum of the squares (SRSS) is the method that can be simplified as the sum of the first member and half-sum of all other members, in cases when the first addend is dominant. A half of the participating load of an individual higher mode is approximately equal to the total load belonging to the spectrum whose acceleration is $a_{gr}/2$ [17], and so the corresponding combination can approximately be described with the following expression:

$$\sum F_{i} = \pm m_{ei} a_{si} \left(a_{gr} \right) F_{fi} / |F_{fi}| + \sum_{i=2}^{n} \pm m_{ei} a_{si} \left(a_{gr} / 2 \right) F_{fi} / |F_{fi}|$$
(5)

The remaining part of the procedure is analogous to that used for the linear combination.

3. Examples

Two examples of spatial RC frames, i.e. the 5-storey frame and 9-storey frame, are analysed in this section. The analysis is carried out with the following assumptions and limitations: (i) analysis of dynamic properties (ϕ_r , T_r , m_e) is carried out on the linear model of the frame with floor slabs and the resulting elastic modulus E_{σ} (ii) the analysis does not include the accidental torsion effect; (iii) frames from both examples are symmetrical in both directions.

3.1. Example 1: 5-storey spatial RC frame

3.1.1. Model description and load vector analysis

The 5-storey spatial RC frame with dimensions as shown in Figure 2a is analysed. Beam dimensions and discretisation of their cross-sections are shown in Figures 2d and 2e, while columns are presented in Figure 2f. Cross-sectional properties are allocated to the frame as shown in Figure 2c. The load distribution is assigned across beams as shown in Figure 2b, while concentrated forces are assigned in nodes at all beam and columns intercepts. Concentrated masses (cf. Figure 2b) and the material modulus E=24.38 GPa as the resulting modulus on the concrete model (cf. Figure 2h), are defined for the analysis of eigenvectors. The numerical model of concrete and reinforcing steel presented in Figures 2g and 2h is used for the analysis of the bearing capacity



Figure 2. Example 1: space frame; b) load on beams and concentrated masses; c) cross-section marks; d)-f) cross-section discretisation; g) reinforcing-steel model; h) concrete model

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Vector	<i>T</i> ;[s]	m _{ei} [%]	<i>m_{ei}</i> [t]	F _{fi} [kN]
$\Phi_{_{\mathcal{I}}}$	0,905	83,12	701,2	1694
$\Phi_{_2}$	0,281	8,65	73,0	1813
$\Phi_{_{\mathcal{J}}}$	0,168	3,93	33,1	2752
$\Phi_{_4}$	0,118	0,69	5,8	1712
Φ_5	0,113	0,59	5,0	1955





Figure 3. Bearing capacity curve for vector ϕ_1 (direction x), and the cross-section plastification order



Figure 4. Seismic resistance analysis for the first vector (F1)



Figure 5. Bearing capacity curve for linear combination of modes and x direction

curve. The total weight of the model is W=8280 kN , and the total mass of the model is M=844.0 tons.

Eigenvectors ϕ_{i} , and the data on the corresponding periods T_i and participating masses m_{ei} are obtained through resolution of the linear dynamic task, as shown in Table 1.

The distribution and intensity of lateral horizontal forces on the frame are determined according to the expression (1). The resulting vector of horizontal load is monotonously increased in the nonlinear analysis until failure, i.e. until the critical load factor is achieved. The sum of all horizontal forces results in the failure force F_{f} for each individual eigenvector.

The form of the first load vector in the limit state, the corresponding bearing capacity curve, and the order of occurrence of plastification hinges, are presented in Figure 3 for the direction x. The early start of plastification is due to the realization of load in the way typical for the pushover method, where the vertical gravity load is imposed in the first phase, and the horizontal seismic load in the second phase. Therefore, the plastification starts immediately after the start of the second loading phase.

The conversion of the bearing capacity curve into the ADRS format, determination of elastoplastic equivalent, and limit target ground acceleration according to EN 1998, are shown in Figure 4 for the first mode and for the direction x.

3.1.2. Determination of target acceleration for x direction

During determination of target acceleration, the target acceleration obtained for the first vector, i.e. $a_{ar.1}$ = 0.43 g, is assumed in the first step as the starting acceleration. a similar procedure is made for the remaining modes. The trial method procedure, with the result for minimum a_{qr_i} value for the case of target acceleration of linear combination of modes for x direction, is presented in Table 2. The target acceleration obtained amounts to $a_{qr,t}$ = 0.205 g. The corresponding force in the cross-section at the base of the calculation model (2188 kN) is not the smallest failure force.

a _{gr,i}	a _{s,1}	F,	a _{s,2}	F_2	а _{s,3}	F3	a _{s,4}	F_4	а _{s,5}	F ₅	ΣF_i	F_{L}	$F_L / \Sigma F_i$	δ_1	δ_{z}	$\delta_{_{\mathcal{J}}}$	δ_4	$\delta_{\mathfrak{s}}$
0,43	0,247	1694	1,272	911	1,425	463	0,780	44,5	0,780	38,2	3156	2241	0,71	0,574	2,958	3,314	1,814	1,814
0,25	0,224	1541	0,783	561	0,938	305	0,610	34,8	0,610	29,9	2471	2051	0,83	0,896	3,132	3,752	2,440	2,440
0,21	0,207	1424	0,678	485	0,788	256	0,510	29,1	0,510	25,0	2220	2175	0,98	0,986	3,229	3,752	2,429	2,429
0,205	0,205	1410	0,664	476	0,769	250	0,492	28,0	0,492	24,1	2188	2188	1,00	1,000	3,239	3,751	2,400	2,400
0,20	0,202	1390	0,652	467	0,750	244	0,483	27,5	0,483	23,7	2151	2194	1,02	1,010	3,260	3,750	2,415	2,415
0,18	0,189	1300	0,585	419	0,675	219	0,443	25,3	0,443	21,7	1985	2204	1,11	1,050	3,250	3,750	2,461	2,461

Table 2. Target acceleration at linear combination of modes for x direction

	Table 3	3. Tar	get acc	eleration	for th	e root	(SRSS)	combination	of	modes	and f	or	x direction
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a _{gr,i}	a _{s,1}	F,	a _{gr,i} /2	a	<i>F</i> ₂	а _{s,3}	F3	a _{s,4}	F ₄	a _{s,5}	<i>F</i> ₅	ΣF_i	F _{SRSS}	$F_L / \Sigma F_i$	δ_1	δ_2	$\delta_{_{\mathcal{B}}}$	δ_4	δ_5
0,43	0,247	1699	0,215	0,678	485	0,806	262	0,510	29,1	0,510	25,0	2501	2201	0,88	0,574	1,577	1,874	1,186	1,186
0,27	0,233	1603	0,135	0,426	305	0,506	165	0,315	18,0	0,315	15,4	2106	2106	1,00	0,863	1,578	1,874	1,167	1,167
0,25	0,224	1541	0,125	0,400	286	0,469	152	0,285	16,2	0,285	14,0	2010	2050	1,02	0,896	1,600	1,876	1,140	1,140
0,20	0,202	1390	0,100	0,332	238	0,375	122	0,233	13,3	0,233	11,4	1774	2040	1,15	1,010	1,660	1,875	1,165	1,165

Load vectors for all participating models, bearing capacity curves, and the order of occurrence of plastification hinges for linear combination of modes and for x direction, are presented in Figure 5.

Results for the root (SRSS) combination of modes and for x direction are presented in Table 3. The target acceleration amounts to $a_{gr,t} = 0.27$ g. It can be concluded by comparing target accelerations that the linear combination of modes is less favourable than the root combination (SRSS) as the structural failure occurs at the lower target acceleration ($a_{gr,t} = 0.205$ g), unlike the target acceleration for the SRSS combination ($a_{ar,t} = 0.27$ g).

Load vectors for participating modes, bearing capacity curves, and the order of occurrence of plastification hinges for the root (SRSS) combination of modes and for x direction, are presented in Figure 6.



Figure 6. Bearing capacity curve for the root (SRSS) combination of modes and for x direction

3.1.3. Determination of target acceleration for y direction

Analysis results for y direction are presented below. Load vector properties for each individual mode are presented in Table 4.

Table 4. Horizontal load vector properties for y direction

Vector	<i>T</i> ,[s]	m _{ei} [%]	<i>m_{ei}</i> [t]	$F_{_{fi}}[kN]$
$\Phi_{_{1}}$	1,002	81,97	691,8	1418
Φ_2	0,317	10,43	88,0	1563
$\Phi_{_{\mathcal{J}}}$	0,176	4,14	34,9	2436
$\Phi_{_4}$	0,119	1,92	16,2	2225
$\Phi_{_{5}}$	0,094	0,63	5,3	1180

Determination of target acceleration for the linear combination of modes for y direction is shown in Table 5. The target acceleration obtained for the first vector, i.e. $a_{gr,\tau} = 0.52$ g, is taken to be the initial acceleration in the first step. The target acceleration obtained amounts to $a_{art} = 0.17$ g.

Figure 7 shows load vectors of all participating modes, bearing capacity curves, and the order of occurrence of plastification hinges for the linear combination of modes and for y direction.

Definition of target acceleration for the SRSS combination of modes and for y direction is shown in Table 6. The

Target acceleration in multimodal pushover method for R/C frames

a _{gr,i}	а _{s,1}	<i>F</i> ₁	a _{s,2}	<i>F</i> ₂	а _{s,3}	F3	a	F_4	a _{s,5}	F ₅	ΣF_i	F	$F_L / \Sigma F_i$	$\delta_{_{1}}$	δ_2	$\delta_{_{\mathcal{B}}}$	$\delta_{_4}$	$\delta_{_{5}}$
0,52	0,209	1418	1,355	1169	1,950	669	0,975	155	0,765	39,8	3451	2105	0,61	0,402	2,606	3,750	1,875	1,471
0,34	0,195	1323	0,976	842	1,275	437	0,638	101	0,510	26,5	2731	2021	0,74	0,574	2,871	3,750	1,876	1,500
0,20	0,183	1242	0,631	545	0,750	257	0,383	61	0,300	15,6	2120	1929	0,91	0,915	3,155	3,750	1,915	1,500
0,17	0,166	1126	0,535	462	0,638	219	0,525	84	0,413	21,4	1912	1912	1,00	0,976	3,147	3,753	3,088	2,429
0,15	0,153	1038	0,495	427	0,563	193	0,495	79	0,390	20,3	1757	1863	1,06	1,020	3,300	3,753	3,300	2,600

Table 5. Target acceleration at linear combination of modes for y direction

Table 6. Target acceleration for the root (SRSS) combination of modes and for y direction

a _{gr,i}	a _{s,1}	F,	a _{gr,i} /2	a _{s,2}	F_2	а _{s,3}	F3	a _{s,4}	F_4	a _{s,5}	F_{5}	ΣF_i	F _{srss}	$F_{SRSS}/\Sigma F_{i}$	$\delta_{_{1}}$	δ_2	$\delta_{_{\mathcal{J}}}$	δ_4	$\delta_{_{5}}$
0,52	0,209	1418	0,26	0,750	647	0,975	334	0,443	70,4	0,342	17,8	2488	1965	0,79	0,402	1,442	1,875	0,852	0,658
0,34	0,195	1323	0,17	0,530	457	0,638	219	0,285	45,3	0,225	11,7	2056	1871	0,91	0,574	1,559	1,876	0,838	0,662
0,22	0,189	1282	0,11	0,389	336	0,413	142	0,165	26,2	0,158	8,2	1794	1758	0,98	0,859	1,768	1,877	0,750	0,718
0,21	0,186	1262	0,105	0,380	328	0,394	135	0,150	23,9	0,147	7,6	1756	1757	1,00	0,886	1,810	1,876	0,714	0,700
0,20	0,183	1242	0,10	0,347	300	0,375	129	0,135	21,5	0,130	6,8	1698	1732	1,02	0,915	1,735	1,875	0,675	0,650



Figure 7. Bearing capacity curve for linear combination of modes for y direction



Figure 8. Bearing capacity curve for the root (SRSS) combination of modes and for y direction

target acceleration amounts to $a_{gr,t} = 0.21$ g. It was established that the linear combination is relevant for the direction y as well, because the structural failure occurs at a lower target acceleration ($a_{gr,t} = 0.17$ g), unlike the target acceleration for the root combination ($a_{gr,t} = 0.21$ g). Figure 8 shows load vectors of participating modes, bearing capacity curves, and the order of occurrence of plastification hinges for the SRSS combination of modes and for y direction.

3.2. Example 2: 9-storey spatial RC frame

3.2.1. Model description and load vector analysis

The 9-storey spatial RC frame with dimensions as shown in Figure 9a is analysed. Beam dimensions and discretisation of their cross-sections are shown in Figures 9d and 9e, while columns are presented in Figure 9f. Cross-sectional properties are allocated to the frame as shown in Figure 9c. The load distribution is assigned across beams as shown in Figure 9b, while concentrated forces are assigned in



Figure 9. Example 2: a) space frame; b) load on girders and concentrated masses; c) cross-section marks; d)-f) cross-section discretisation; g) reinforcing-steel model; h) concrete model

nodes at all beam and columns intercepts. Concentrated masses (cf. Figure 9b) and the material modulus E=24.38 GPa as the resulting modulus on the concrete model (cf. Figure 9h), are defined for the analysis of eigenvectors. The numerical model of concrete and reinforcing steel presented in Figures 9g and 9h is used for the analysis of the bearing capacity curve. The total weight of the model is W=14903 kN , and the total mass of the model is M=1519.2 tons.

Eigenvectors ϕ_{μ} and the data on the corresponding periods T_i and participating masses m_{ei} are obtained through resolution of the linear dynamic task, as shown in Table 7.

Vector	<i>T_i</i> [s]	m _{ei} [%]	m _{ei} [t]	F _{fi} [kN]
$\Phi_{_{1}}$	2,298	79,87	1213,4	1330
Φ_2	0,746	10,37	157,5	1530
$\Phi_{_{\mathcal{J}}}$	0,420	3,87	58,8	2035
$\Phi_{_4}$	0,282	2,20	33,4	1770
$\Phi_{_5}$	0,207	1,38	21,0	2490

Table 7. Horizonta	l load vector	properties in	x direction
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3.2.2. Determination of target acceleration for x direction

The determination of target acceleration of the linear combination of modes for the direction x is shown in Table 8. The target acceleration obtained for the first vector, amounting to $a_{gr,i} = 0.92$ g, is assumed to be the initial acceleration in the first step. The target acceleration obtained amounts to $a_{art} = 0.19$ g.

Load vectors for all participating models, bearing capacity curves, and the order of occurrence of plastification hinges for linear combination of modes and for x direction, are presented in Figure 10.

Results for the root (SRSS) combination of modes and for x direction are presented in Table 9. The target acceleration amounts to $a_{grt} = 0.25$ g. It can be concluded by comparing target accelerations that the linear combination of modes for the 9-storey frame is significantly less favourable than the root combination (SRSS) because the structural failure occurs at lower target acceleration ($a_{grt} = 0.19$ g), unlike the target acceleration for the SRSS combination ($a_{grt} = 0.257$ g), which is the same case as in the 5-storey frame.

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a _{gr,i}	a _{s,1}	F,	a _{s,2}	<i>F</i> ₂	a _{s,3}	F3	a _{s,4}	F_4	a _{s,5}	F_{5}	ΣF_i	F_{L}	$F_L / \Sigma F_i$	δ_{τ}	δ_2	$\delta_{_{\mathcal{F}}}$	δ_4	δ_5
0,92	0,112	1330	0,923	1426	2,239	1292	2,700	886	3,161	651	5585	2681	0,48	0,121	1,003	2,434	2,935	3,436
0,50	0,107	1274	0,694	1072	1,428	824	1,749	574	1,858	383	4126	2476	0,60	0,214	1,388	2,856	3,498	3,716
0,20	0,073	869	0,448	692	0,689	398	0,750	246	0,750	155	2359	2123	0,90	0,365	2,240	3,445	3,750	3,750
0,19	0,071	845	0,318	491	0,660	381	0,713	234	0,713	147	2098	2098	1,00	0,374	1,674	3,474	3,753	3,753
0,18	0,068	809	0,307	474	0,623	359	0,675	221	0,675	139	2004	2164	1,08	0,378	1,706	3,461	3,750	3,750

Table 8. Target acceleration at linear combination of modes for the x direction

Table 9. Target acceleration for the root (SRSS) combination of modes and for x direction

a _{gr,i}	а _{s,1}	F,	a _{gr,i} /2	а _{s,2}	<i>F</i> ₂	а _{s,3}	F,	а _{s,4}	F_4	а _{,,5}	F ₅	ΣF_i	F _{srss}	$F_{SRSS}/\Sigma F_{i}$	$\delta_{_{1}}$	δ_2	$\delta_{_{\mathcal{J}}}$	δ_4	$\delta_{_{5}}$
0,92	0,112	1330	0,46	0,651	1006	1,350	779	1,637	537	1,725	355	4010	2085	0,52	0,122	0,708	1,467	1,779	1,875
0,50	0,107	1274	0,25	0,391	604	0,845	488	0,938	308	0,938	193	2866	2064	0,72	0,214	0,782	1,690	1,876	1,876
0,30	0,093	1107	0,15	0,260	402	0,523	302	0,538	176	0,563	116	2103	2439	1,16	0,310	0,867	1,743	1,793	1,877
0,25	0,085	1011	0,125	0,225	348	0,442	255	0,469	154	0,469	97	1865	1865	1,00	0,340	0,900	1,768	1,876	1,876
0,20	0,073	869	0,10	0,182	281	0,354	204	0,375	123	0,375	77	1555	1368	0,88	0,365	0,910	1,770	1,875	1,875



Figure 10. Bearing capacity curve for linear combination of modes for x direction



Figure 11. Bearing capacity curve for the root (SRSS) combination of modes and for *x* direction

Load vectors for all participating modes, bearing capacity curves, and the order of occurrence of plastification hinges for the root (SRSS) combination of modes and for x direction, are presented in Figure 11.

3.2.3. Determination of target acceleration for y direction

Analysis results for *y* direction are presented below. Load vector properties for each individual mode are presented in Table 10.

Table 10. Horizontal load vector properties for y direction

Vector	<i>T</i> ;[s]	m _{ei} [%]	m _{ei} [t]	$F_{ff}[kN]$
$\Phi_{_1}$	2,715	78,91	1198,8	1410
Φ_2	0,869	10,59	160,9	1735
$\Phi_{_{\mathcal{F}}}$	0,478	4,10	62,3	1965
Φ_4	0,317	1,60	24,3	2340
Φ_{5}	0,224	1,40	21,3	2435

Determination of target acceleration for the linear combination of modes for y direction is shown in Table 11. The target acceleration obtained for the first vector, i.e. $a_{qr,i} = 1.05$ g, is taken to be the initial acceleration in the

a _{gr,i}	а _{s,1}	F,	a _{s,2}	<i>F</i> ₂	a _{s,3}	F3	a _{s,4}	F_4	a _{s,5}	F_{5}	ΣF_i	F_{L}	$F_L / \Sigma F_i$	δ_{τ}	δ_2	δ_3	δ_4	δ_5
1,05	0,120	1410	0,917	1447	2,313	1413	3,900	928	3,606	754	5952	2619	0,44	0,114	0,873	2,203	3,714	3,434
0,40	0,097	1141	0,494	780	1,063	649	1,500	357	1,500	314	3240	2203	0,68	0,243	1,235	2,658	3,750	3,750
0,20	0,063	741	0,309	488	0,588	359	0,750	179	0,750	157	1923	1846	0,96	0,315	1,545	2,940	3,750	3,750
0,18	0,058	682	0,277	437	0,537	328	0,675	161	0,675	141	1749	1749	1,00	0,322	1,539	2,983	3,750	3,750
0,17	0,056	659	0,265	418	0,500	306	0,638	152	0,638	133	1667	1834	1,10	0,329	1,559	2,941	3,753	3,753

Table 11. Target acceleration at linear combination of modes for y direction



Figure 12. Bearing capacity curve for linear combination of modes for y direction



Figure 13. Bearing capacity curve for the root (SRSS) combination of modes for y direction

Table 12. Target acceleration for the root (SRSS) combination of modes and for y direction

first step. The target acceleration obtained amounts to a_{ort} = 0.18 g.

Figure 12 shows load vectors of all participating modes, bearing capacity curves, and the order of occurrence of plastification hinges for the linear combination of modes and for y direction. Definition of target acceleration for the root (SRSS) combination of modes and for y direction is shown in Table 12. The target acceleration amounts to a_{art} = 0.245 g.

It was established that the linear combination of modes for the 9-storey frame is relevant for the direction y as well, because the structural failure occurs at a lower target acceleration ($a_{gr,t} = 0.18$ g), unlike the target acceleration for the root (SRSS) combination ($a_{gr,t} = 0.245$ g), which is the same case as for the 5-storey frame. Figure 13 shows load vectors of participating modes, bearing capacity curve and the order of occurrence

curve, and the order of occurrence of plastification hinges for the SRSS combination of modes and for y direction.

4. Test of frequency of the linear (L) and root (SRSS) mode combinations

The objective of the following procedure is to check which of the two mode combinations is more probable based

a _{gr,i}	a _{s,1}	F,	a _{gr,i} /2	a _{s,2}	<i>F</i> ₂	а _{s,3}	F3	a _{s,4}	F_4	a _{s,5}	F_{5}	ΣF_i	F _{srss}	$F_{SRSS}/\Sigma F_{i}$	δ_1	δ_{z}	$\delta_{_{\mathcal{J}}}$	δ_4	$\delta_{\mathfrak{s}}$
1,05	0,120	1410	0,525	0,610	963	1,300	794	1,968	468	1,879	393	4028	2175	0,54	0,114	0,581	1,238	1,874	1,790
0,40	0,097	1141	0,20	0,309	488	0,588	359	0,750	179	0,750	157	2323	1812	0,78	0,243	0,773	1,470	1,875	1,875
0,245	0,073	858	0,123	0,207	327	0,357	218	0,459	109	0,459	96	1608	1608	1,00	0,298	0,845	1,457	1,873	1,873
0,24	0,072	847	0,12	0,203	320	0,350	214	0,450	107	0,450	94	1582	1661	1,05	0,300	0,846	1,458	1,875	1,875
0,20	0,063	741	0,10	0,173	273	0,295	180	0,375	89	0,375	78	1362	1702	1,25	0,315	0,865	1,475	1,875	1,875

on verification on a set of the following five selected realearthquake accelerograms: Montenegro – 1979, Campano Lucano (Italia) – 1980, Aigion (Greece) – 1995, Strofades (Greece) – 1997 and Olfus (Iceland) – 2008, as shown in Figure 14. Accelerogram records were taken from [18].

The verification is conducted on the linear and nonlinear levels for single-degree-of-freedom systems (SDOF) taken as modes obtained for RC frame models from examples 1 and 2. The dynamic response for each SDOF was obtained using the Newmark's average acceleration method. The numerical integration was obtained using the computer program written in C++ according to the procedure presented in [19].



Figure 14. Accelerograms from five earthquake, [18]

4.1. Verification of linear and root mode combinations on linear level

First, the 5-storey RC space frame from example 1 is analysed. Transverse forces in cross-section were calculated based on the design model (F_i) as time functions for each of the five eigenmodes. At that, the stiffness k of the linear SDOF was defined via the known period T_i and the corresponding participating mass m_r , which corresponds to an individual eigenvector ϕ_r .

A relative force scale was introduced. According to this scale, the biggest force in all modes ($|F_{max}|$) is equal to 1.00, and the value of other forces is determined in relation to this value.

Figure 15 shows the diagram of the transverse force in cross-section based on the design model (*F*) dependent on time for all five eigenmodes for the example of the 5-storey RC framed subjected to load in x direction through the Montenegro earthquake seismic excitation. It can be observed that, on the relative force scale, the biggest force in all five modes (|*F*_{max}|=1135 kN) corresponds to the value of 1.00.



Figure 15. Transverse forces in cross section based on the design model for the 5-storey RC frame for all five modes, dependent on time, for the Montenegro seismic excitation in x direction

Maximum values of transverse forces in cross section based on calculation model for all modes shown in tables 13 to 16, due to seismic action of five selected earthquake records, were obtained analogously.

Analysis results for the directions x and y, given in Table 13, show simultaneously the least favourable combination of the linear superposition of modes F_{μ} and the least favourable combination of the square root of the sum of the squares F_{SRSS} . A similar procedure was conducted for the analysis of the 9-storey linear RC space frame from example 2, and the analysis results are given in Table 14.

Dynamic response results for the linear system, as given in tables 13 and 14, show that the linear mode combination F_L is more often unfavourable than the root combination of the sum of the squares F_{spss} .

Table 13. Comparison of linear (L) and root (SRSS) mode combinations for the 5-storey RC frame

Table 14. Comparison of linear (L) and root (SRSS) mode combinations for the 9-storey RC frame

Monte	enegro:	5-stor	ey RC f	rame, x	direction	Mont	Montenegro: 5-storey RC frame, y direction								
Finar	F 2max	F _{3max}	F dmax	F Smar	at the same	e time t	F total	F 2max	F 3max	Fanar	F Smar	at the same	e time t		
0.72	0.47	0.25	0.03	0.03			1.00	0.62	0.59	0.31	0.03	01010 1011			
F tmin	F 2min	F 3min	Famin	Famin	FL	-0.98	Finin	F _{2min}	F 3min	F _{dmin}	F Smin	FL	-1.31		
-1.00	-0.34	-0.31	-0.03	-0.03	F 3853	1.09	-1.00	-0.64	-0.59	-0.32	-0.03	F SASS	1.13		
Camp	ano Lu	cano: 5	-storey	RC fra	me, x direc	tion	Campano Lucano: 5-storey RC frame, y direction								
F smar	tmax F 2max F 3max F 4max F 5max with a comparison t				Finar	F 2mar	Famar	Famar	Famar	at the same ti					
1.00	0.38	0.29	0.02	0.02	at the same	eumer	0.97	0.66	0.31	0.21	0.01	at the same	e ume t		
Ftmin	F Junio	F _{3min}	Famin	Famin	FL	0.91	Ftmin	F _{2min}	F 3min	F dmin	F Smin	FL	1.10		
-0.99	-0.51	-0.30	-0.02	-0.03	Fanss	1.00	-1.00	-0.77	-0.35	-0.19	-0.02	F 3433	1.01		
Aigion	: 5-sto	rey RC	frame,	x direc	tion		Aigion: S-storey RC frame, y direction								
Finne	Fame	Fame	FARM	Form			Finn	Farmer	Fame	Firm	Farm				
1.00	0.94	0.59	0.03	0.03	at the same	eumer	0.91	0.70	0.87	0.13	0.02	at the same	e ume t		
Frain	F Junio	F Junio	Ferrir	Famin	FL	-1.90	Frain	F _{2min}	F 3min	F denie	Famin	FL	-1.59		
-0.94	-1.00	-0.61	-0.03	-0.04	F 1811	1.25	-1.00	-0.70	-0.80	-0.14	-0.03	F 1811	1.19		
Strofa	des: 5-	storey	RC fran	ne, x di	rection		Strofa	ides: 5-	storey	RC fra	ne, y di	rection			
Finar	Famer	Fame	Famer	Former	at the case	a time t	Finar	F 2mar	Famer	Finar	Famer	at the case	. time t		
1.00	0.46	0.21	0.02	0.02	at the same	e ume t	1.00	0.48	0.25	0.09	0.01	at the same	e ome c		
F tmin	F Jonie	F Junio	Famin	Famin	FL	-0.97	Finin	F _{2min}	F 3min	F denie	Famin	FL	-1.02		
-0.97	-0.50	-0.22	-0.01	-0.02	F 3833	1.00	-0.92	-0.48	-0.26	-0.10	-0.01	F 3833	1.00		
Olfus:	5-stor	ey RC f	rame, x	directi	ion		Olfus	5-stor	ey RC f	rame, y	directi	on			
Famer	Famor	Famue	Famer	Fores	at the same	e time t	Finan	Famor	Famar	Fanar	Former	at the same	e time t		
0.99	0.69	0.45	0.07	0.06			1.00	0.54	0.40	0.23	0.02	and and plant			
F tmin	F Junio	F Junio	Famin	Famin	FL	1.15	Finin	F _{2min}	F 3min	F denie	Famin	FL	1.26		
-1.00	-0.66	-0.43	-0.08	-0.06	Faess	1.04	-0.91	-0.47	-0.36	-0.22	-0.02	Farss	1.05		

Table 15. Linear (L) and root (SRSS) combination of nonlinear model for the 5-storey RC frame

Mont	enegro	5-stor	ey RC f	frame, i	direction		Monte	enegro:	5-stor	ey RC f	rame, y	direction	
F 1000	F 2mar 0.58	F 3max 0.67	F 4mer 0.10	F Smar	at the sam	e time t	F 100	F 2max 0.62	F 3mar 0.59	Famar 0.31	F 5max 0.03	at the same	etime
Freeh	F _{2min}	Famire	Famin	Famin	F.	-1.89	Finit	F 2min	F _{3min}	Family	Famin	F.	1.3
-1.00	-0.56	-0.66	-0.09	-0.03	F SRSS	1.18	-1.00	-0.64	-0.59	-0.32	-0.03	F 5455	1.0
Camp	ano Lu	cano: 5	-store	RC fra	me, x direc	ction	Camp	ano Luc	cano: 5-	-storey	RC fra	me, y direc	tion
F team 1.00	F 3mar 0.38	F 3max 0.29	F 4max 0.02	F 5max 0.02	at the sam	e time t	F tmax 0.97	F 2max 0.66	F 3mar 0.31	F denas 0.21	F tmax 0.01	at the same	e time
Ftmin	F 2min	F Junio	Famin	Famin	FL	1.26	Finis	F _{2min}	Famin	Famin	Famin	FL	1.7
-0.99	-0.51	-0.30	-0.02	-0.03	Fanss	1.01	-1.00	-0.77	-0.35	-0.19	-0.02	F SASS	1.1
Aigior	n S-sto	vrey RC	frame,	x direc	tion		Aigion	S-sto	rey RC	frame,	y direct	tion	
F team 1.00	F 2mar 0.94	F 3max 0.59	F 4max 0.03	F 5mar 0.03	at the sam	e time t	F tease 0.91	F 2max 0.70	F 3max 0.87	F 4max 0.13	F 5mar 0.02	at the same	e time
Frain	Famin	Famin	Famin	Famin	FL	-1.55	Finis	Famin	Famin	Famin	Famin	FL	1.7
-0.94	-1.00	-0.61	-0.03	-0.04	Fana	1.05	-1.00	-0.70	-0.80	-0.14	-0.03	F 5811	1.10
Strofa	ides: 5	storey	RC fra	me, x d	irection		Strofa	des: 5-	storey	RC fran	me, y di	rection	
F 1.00	F 2mar 0.46	F 3mar 0.21	F 4mar 0.02	F 5mar 0.02	at the sam	e time t	F tmax 1.00	F 2max 0.48	F 3mar 0.25	F 4max 0.09	F	at the same	e time
Ftmin	F 2min	Famin	Famin	F sain	FL.	1.38	F tonio	Famin	F 3min	Famin	Famin	FL	1.2
-0.97	-0.50	-0.22	-0.01	-0.02	FSRSS	1.09	-0.92	-0.48	-0.26	-0.10	-0.01	F 5455	1.0
Olfus	5-stor	ey RC f	rame, >	c direct	ion		Olfus:	5-stor	ey RC fi	rame, y	directi	on	
F 1mar 0.99	F 3mar 0.69	F 3max 0.45	F 4mar 0.07	F 5max 0.06	at the sam	e time t	F tmax 1.00	F 2max 0.54	F 3mar 0.40	F denar 0.23	F 5mar 0.02	at the same	e time
Ftmin	F 2min	F 3min	Fanin	F tonin	FL	1.48	Funie	F _{2min}	F _{3min}	F denie	Finin	FL	-1.5
-1.00	-0.66	-0.43	-0.08	-0.06	France	1.13	-0.91	-0.47	-0.36	-0.22	-0.02	Freeze	1.0

4.2. Verification of linear and root mode combinations on nonlinear level

The nonlinear level is realized by treating each mode as a single nonlinear SDOF. Nonlinear cyclic properties of the SDOF are generated from the corresponding bearing capacity curve as shown in Figure 16 and relation (6).



Figure 16. Cyclic bearing capacity curve for nonlinear system

Monte	enegro:	9-stor	ey RC f	rame, »	direction		
Famer	Famer	Famer	Famar	Former	stille com		
0.83	0.43	0.49	0.36	0.21	at the same	e cime c	
Frain	Famin	F 3min	Famin	Famin	FL.	-1.45	
-1.00	-0.59	-0.46	-0.27	-0.26	F 1953	1.14	
Camp	ano Lu	cano: 9	-storey	RC fra	me, x dire	ction	
Finar	Famer	Famer	Famer	Finar			
1.00	0.63	0.50	0.30	0.25	at the sam	e time t	
Ftmin	F _{2min}	F _{3min}	Famin	F tenin	FL	-1.36	
-0.87	-0.47	-0.60	-0.25	-0.25	FSRSS	1.14	
Aigion	: 9-sto	rey RC	frame,	x direc	tion		
Finan	Farmer	Famer	Farmer	Finan			
0.90	0.52	0.51	0.84	0.40	at the sam	etmet	
Frain	Finin	F 3min	Famin	Formin	FL	-1.59	
-1.00	-0.53	-0.59	-0.99	-0.35	F 1833	1.15	
Strofa	des: 9-	storey	RC fran	me, x di	irection		
Finar	F 2mar	F 3mar	Famar	Filmar			
1.00	0.47	0.21	0.13	0.09	at the sam	e time t	
Frain	F 2min	F José	Famin	Famin	FL	-1.13	
-0.98	-0.58	-0.20	-0.13	-0.09	Farss	1.02	
Olfus:	9-stor	ey RC f	rame, x	directi	ion		
Finan	Famer	Famer	Ferrer	Former			
1.00	0.20	0.25	0.17	0.24	at the sam	e time t	
Finin	Finin	F 3min	Famin	F Smin	FL	0.97	
-0.92	-0.17	-0.22	-0.15	-0.21	From	1.00	

Montenegro: 9-storey RC frame, y direction												
Fine	Famer	Famer	FARME	Fame								
0.71	0.25	0.58	0.16	0.22	at the sam	e cime t						
Finis	Famin	F Junio	F Amin	Famin	F.	-1.06						
-1.00	-0.39	-0.65	-0.14	-0.21	Fanss	1.00						
Campano Lucano: 9-storey RC frame, y direction												
F tonas F 2max F 2max F 4max F 5max at the same time t												
0.90	1.00	0.58	0.18	0.23	at the sam	e time t						
Finis	F _{2min}	F _{3min}	Famin	F Smin	FL	1.72						
-0.95	-0.80	-0.72	-0.25	-0.21	Farss	1.11						
Aigion: 9-storey RC frame, y direction												
Fina	F Jones	F James	Finer	Farmer								
0.97	0.32	0.39	0.45	0.27	at the sam	e time t						
Famin	F praise	F Join	Fanin	Formin	FL	-1.35						
-1.00	-0.33	-0.40	-0.50	-0.25	F 1833	1.03						
Strofa	des: 9-	storey	RC fra	me, y di	irection							
Ftmar	F 2max	F 3max	Famar	Finar								
0.79	0.80	0.25	0.18	0.19	at the sam	e time t						
Funit	F 2min	F _{3min}	Famin	Famin	FL	1.16						
-1.00	-0.90	-0.28	-0.18	-0.18	F SASS	1.09						
Olfus:	9-stor	ey RC f	rame, y	direct	ion							
Ftmax	F 2max	F Jmax	Fdmax	Former	at the care	e time t						
0.94	0.29	0.32	0.20	0.41	at the sam	e unhe t						
Finie	F 2min	F Join	F _{énin}	F Smin	FL	1.06						
-1.00	-0.23	-0.33	-0.18	-0.34	Fants	1.00						

Table 16. Linear (L) and root (SRSS) combination of nonlinear model for the O-storey DC frame

		101 0	ine 5	-310	eyne	mann										
Monte	negro:	9-stor	ey RC f	rame, >	direction		Mor	Montenegro: 9-storey RC frame, y direction								
F 1max 0.69	F 2max 0.92	F 3000	F 4max 0.52	F 5max 0.32	at the sam	e time t	F 1m	F 2m	F 3max 4 0.67	F 4max 0.23	F 5mar 0.29	at the sam	e time t			
Finis	F _{2min}	F _{3min}	F damin	Fsmin	FL	-2.02	F to	, F 3m	F F Junio	Famin	Famin	FL	-1.49			
-0.78	-1.00	-0.66	-0.60	-0.34	FIRST	1.31	-1.0	0 -0.6	0 -0.64	-0.24	-0.33	Farss	1.13			
Campa	ino Luc	ano: 9	-storey	RC fra	me, x direc	tion	Campano Lucano: 9-storey RC frame, y direction									
Frank Frank Frank Frank Frank							Fim	F 2m	F Famer	Famer	Famer					
0.73	0.95	0.67	0.33	0.27	at the sam	e time t	0.9	0.9	0.71	0.15	0.33	at the same	e time t			
Finin	Finin	F _{3min}	Famin	Formin	FL	-1.68	F .m	, F 2m	n Fanin	Famin	Famin	FL	-2.14			
-0.68	-1.00	-0.65	-0.36	-0.29	Fansa	1.32	-1.0	0 -0.8	7 -0.68	-0.18	-0.36	F 5833	1.34			
Algion: 9-storey RC frame, x direction								in: 9-st	torey RC	frame,	y direc	tion				
Fina	Fanar	Fanne	Farmer	Finar			Fim	F 200	F Fame	Farmer	Fame					
0.89	0.92	0.99	1.00	0.85	at the sam	e time t	0.7	0.6	3 0.89	0.13	0.42	at the sam	e time t			
Finin	F _{2min}	F _{3min}	Famin	Famin	FL	2.12	F	F 2m	F Junio	Famin	Finin	FL	1.93			
-0.98	-0.89	-0.99	-0.85	-0.93	Fanss	1.41	-0.7	5 -0.5	8 -1.00	-0.14	-0.47	Fanss	1.13			
Strofa	des: 9-	storey	RC fran	me, x di	rection		Stro	ades:	9-storey	RC fra	me, y đi	irection				
Finan	Fame	Fame	Farmer	Finar			Fin	Fam	F	Farmer	Fame					
0.83	0.69	0.62	0.21	0.18	at the sam	e time t	0.9	0.8	3 0.68	0.16	0.33	at the sam	e time t			
Ftmin	F _{2min}	F _{3min}	Famin	Famin	FL	1.37	F	F 3m	F Junio	Famin	Finin	FL	1.78			
-1.00	-0.75	-0.60	-0.21	-0.19	Fanss	1.19	-1.0	0 -0.9	2 -0.67	-0.11	-0.32	Fans	1.16			
Olfus:	9-stor	ey RC f	rame, x	directi	ion		Olfu	s: 9-st	orey RC f	rame, y	direct	ion				
Finar	Famer	Fame	Fdrag	Famer	at the case	e time t	Fim	F 2m	Famer	Famer	Finan	at the case	a time t			
0.96	0.71	1.00	0.44	0.25	ar one sam	e unité s	0.9	0.6	5 0.53	0.27	0.24	er, und sam	e unité c			
Ftmin	F _{2min}	F _{3min}	F _{dmin}	F Smie	FL	1.92	F _{1m}	F 3m	F Junio	Fania	Finin	FL	1.53			
-0.97	-0.75	-0.90	-0.38	-0.26	Feats	1.15	-1.0	0 -0.6	5 -0.57	-0.25	-0.25	Fases	1.06			

The functions belonging to the figure are defined by the following analytical form:

$$f_{s1} = A\left(1 - e^{-\frac{u-B}{C}}\right) + D$$

$$f_{s2} = -A\left(1 - e^{\frac{u-B}{C}}\right) + D$$

$$f_{s3} = \frac{A}{C}(u-B) + D$$
(6)

The function f_{s_1} corresponds to the bearing capacity curve for an individual mode. It is used to define coefficients A and Cin expression (6). The function $f_{\rm S3}$ is linear and its inclination corresponds to the initial stiffness of the SDOF. Coefficients B and D are calculated in every time step and are used to define the translation of functions f_{s_1} and f_{s_2} along the axis uand *fs* for coefficients *B* and *D*, respectively.

Analysis results for nonlinear SDOF for the directions x and y, given in Table 15, show simultaneously the least favourable combination of the linear superposition of modes $F_{\!_L\!'}$ and the least favourable root combination of modes $F_{_{SRSS'}}$

A similar procedure was conducted for the analysis of the 9-storey linear RC space frame from example 2, and the analysis results are given in Table 16.

Dynamic response results for the nonlinear system, as given in tables 15 and 16, show that the linear mode combination $F_{\rm L}$ is in all examples much more unfavourable than the root combination of the sum of the squares $F_{\rm SPSC}$.

5. Conclusion

The paper presents the procedure for determining the target acceleration, defined as the smallest base acceleration leading to the limit state of bearing capacity of a structure using the multimodal pushover approach, which represents the lowest seismic resistance. The procedure is based on the repeated nonlinear analysis of the structure by multimodal pushover method and by repeated search for the target acceleration.

The following conclusions can be made based on results obtained on concrete examples:

- The influence of higher modes, the second and third modes in particular, is very significant.
- The target or the least acceleration of the base does not necessarily cause the lowest horizontal failure forces in the cross section at the base of the calculation model. At that, the target acceleration is always lower than the limit acceleration of each mode taken separately.
- It has been demonstrated that the linear (L) combination of modes is more critical due to smaller target acceleration when compared to such acceleration for the root (SRSS) combination of modes.

- The general failure vector belonging to target acceleration differs significantly by form from the form of load of the first vector, but also from the form of load with constant acceleration along the height of the structure, as given in EN 1998.
- The frequency of occurrence of linear combination and root combination of modes, tested on a concrete series of seismic records, on both linear and nonlinear levels, confirms that the linear combination is more probable that the root combination of the sum of the squares.
- The multimodal pushover method a priori starts from the assumption that all relevant modes are actually activated in case of a real-life seismic action.
- Regardless of the method by which modes are combined, the practical use of the multimodal approach shows that the seismic resistance or reliability of the studied RC frame structures is lower than that based on the target displacement criterion as per EN 1998 (Annex B).
- The inclusion of higher modes in examples presented in the paper regularly and significantly reduces the limit peak ground acceleration and this by more than two times in some situations, which is significantly less favourable than the request for reserve in the bearing capacity curve of 150 % as compared to the typical target displacement of the top of the structure.

Acknowledgement

This research was supported by the Ministry of Science, Education and Sport of the Republic of Croatia (research project: "Modelling limit bearing capacity and stability of structures at considerable displacements", No.: 083-0831541-1545). The authors express their gratitude for this support.

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